



STANDARD PRACTICES - ENGINE - INFORMATION - ENGINE - TURBINE NOZZLE ASSEMBLY - NEW
STAGE 1 HPT DUCT SEGMENT RETAINING BOLTS - CATEGORY CODE 7 - MOD.ENG-70-0182

1. Planning Information

A. Effectivity

- (1) Aircraft: Airbus A320
(2) Engine: V2500-A1

B. Reason

To replace the Stage 1 HPT Duct Segment Retaining Bolts with new bolts that have increased length. This will make sure that there is sufficient retention.

C. Compliance

Category Code 7

Accomplish when the supply of superseded parts has been depleted.

D. Approval

The part number transaction shown under the MATERIAL INFORMATION paragraph has been shown to comply with the applicable Federal Aviation Regulations and are FAA-APPROVED for the English Model listed.

E. Reference

- (1) Internal Reference No.
91VA039

F. General Information

The Material Information portion of this Service Bulletin provides the latest information concerning the availability of the Windage Cover Retention Bolts.

G. Material Information

New Part No. (ATA No.)	Qty	Est'd Unit Price (\$)	Keyword	Old Part No. (IPC No.)	Instructions Disposition
2A2474 (72-44-60)	40	12.10	Bolt, Machine	2A0155 (01-040)	(S1)(A)(B)

V2500-ENG-70-0182



H. Instruction/Disposition Code Statements:

(S1) Old and new parts coded (S1) are directly interchangeable.

(A) New Part is currently available.

(B) The old part will continue to be supplied until the supply is used up.

NOTE: The estimated 1995 unit prices shown are provided for planning purposes only and do not constitute a firm quotation. Consult the IAE Price Catalog or contact IAE's Spare Parts Sales Department for information concerning firm prices.