

## International Aero Engines

## **INFORMATION BULLETIN**

STANDARD PRACTICES - ENGINE - INFORMATION - ENGINE - HP TURBINE ROTOR AND STATOR

ASSEMBLY - TO ANNOUNCE THE SPAREABILITY OF A STAGE 2 HPT RETAINING PLATE - CATEGORY

CODE 8 - MOD.ENG-70-0278

#### 1. Planning Information

#### A. Effectivity

(1) Aircraft: Airbus A320

(2) Engine: V2500-A1 Engines

#### B. Reason

To make the Stage 2 HPT Plate available as a spare part for repair of the Stage 2 HPT Blade Assembly.

### C. <u>Compliance</u>

Category Code 8

Accomplish based upon experience with the prior configuration.

#### D. <u>Approval</u>

The part number transactions shown under the Material Information portion of this Bulletin have been shown to comply with the applicable Federal Aviation Regulations and are FAA-APPROVED for the Engine Model listed.

#### E. References

(1) Internal Reference No.

87VA152A-03

### F. General Information

(1) The Material Information portion of this Bulletin provides the latest information concerning the availability of the stage 2 HPT Plate.

#### G. Material Information

New Part No. (ATA No.)	Qty	Est'd Unit Price (\$)	Keyword	Old Part No. (IPC No.)	Instructions Disposition
2A0159 (72-45-32)	72	2.62	Plate-HPT, Stage 2	_ _ (01–013)	(A)

V2500-ENG-70-0278

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## H. <u>Instructions/Disposition Code Statements</u>

(A) New part is now available as a spare for repair of the 2AO4O2 Stage 2 HPT Blade Assembly.

NOTE: The estimated 1994 unit prices shown are provided for planning purposes only and do not constitute a firm quotation. Consult the IAE Price Catalog or contact IAE's Spare Parts Sales Department for information concerning firm prices.