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DATE: Feb.10/06

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V2500-A5/D5 PROPULSION SYSTEMS SERVICE BULLETIN

This document transmits Revision 4 to Service Bulletin EV2500-72-0421 and Revision 1 to the Supplement

<u>Document History</u>

Printed in Great Britain

Service Bulletin Revision Status

Initial Issue Sep.4/02 Initial Issue Jun.23/05

Pavision 1 Oct 1/02

Revision 1 Oct.1/02 Revision 2 Apr.14/03 Revision 3 Jun.23/05

Bulletin Revision 4

Remove Incorporate Reason for change

All pages of the Page 1 and 2 of the To identify HPC stubshafts

Summary Summary that do not require rework.

References updated.

Pages 1 to 14 of the Pages 1 to 17 of the To identify HPC stubshafts

Service Bulletin Service Bulletin that do not require rework.

References updated.

<u>Supplement Revision 1</u>

Remove Incorporate Reason for change

All pages Page 1 To identify HPC stubshafts that do not require rework.

References updated.

V2500-ENG-72-0421
Transmittal - Page 1 of 2

CHECK THAT ALL PREVIOUS TRANSMITTALS HAVE BEEN INCORPORATED If any have not been received please advise Customer Data Services, Rolls-Royce plc, Derby, England © Rolls-Royce plc (date as above) Printed in Great Britain

LIST OF EFFECTIVE PAGES

The effective pages to this Service Bulletin following incorporation of Revision 4 to the Bulletin and Revision 1 to the Supplement are as follows:

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S	upplement		
R	1	1	Feb.10/06



<u>ENGINE - HP COMPRESSOR FRONT STUBSHAFT - INTRODUCTION OF A REVISED HP COMPRESSOR</u> FRONT STUBSHAFT WITH HIGH ENERGY PLASMA COATING

SUMMARY

1. PLANNING

A. EFFECTIVITY

Engine Serial No.

V2500-A5 Prior to V11400

V2500-D5

B. CONCURRENT REQUIREMENTS

None.

C. REASON/PROBLEM

Problem

A change in the coating application process from High Energy Plasma to Standard Energy Plasma, has meant that release of particles from the HP Compressor front stubshaft tungsten carbide cobalt hard coating may occur. This may result in surface spall of the bearing race, which may lead to premature deterioration of the number 3 bearing.

The purpose of this Service Bulletin is to advise Operators of the availability of a new stubshaft part number, coated with High Energy Plasma.

Evidence

The problem has been experienced on engines in service.

Substantiation

The changes introduced by this modification have been the subject of extensive engineering analysis.

Objective

Incorporation of this modification is designed to maintain engine reliability.

D. DESCRIPTION

The changes introduced are a revised HP Compressor front stub shaft which is similar to the existing stub shaft except for the following:

The option to use the standard plasma application process for the tungsten carbide cobalt coating has been deleted.

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The requirement to use fluorescent penetrant inspection of the coating after grinding has been introduced.

The classification of the part has been changed from 'sensitive' to 'sensitive, traceable'.

E. COMPLIANCE

Category Code 7

Accomplish when the supply of superseded parts has been depleted.

F. MANPOWER

In service - Not applicable.

At overhaul - Rework only - 10 hours.

G. INTERCHANGEABILITY OF PARTS

Not affected.

2. MATERIAL INFORMATION

A. PARTS PRICES

Part No.	Description	Unit Price US Dollars
6A8082	Shaft and Seal Assy - Front, HP Compressor	35,090.00
6A8083	Shaft, Balancing and Assy – Front, HP Compressor	30,149.00

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<u>ENGINE - HP COMPRESSOR FRONT STUBSHAFT - INTRODUCTION OF A REVISED HP COMPRESSOR</u> FRONT STUBSHAFT WITH HIGH ENERGY PLASMA COATING

1. Planning Information

A. Effectivity

(1) Airbus A319

V2522-A5, V2524-A5, V2527M-A5 Engines prior to Serial No. V11400

(2) Airbus A320

V2527-A5, V2527E-A5 Engines prior to Serial No. V11400

(3) Airbus A321

V2530-A5, V2533-A5 Engines prior to Serial No. V11400

(4) Boeing - Longbeach Division MD-90

V2525-D5, V2528-D5 Engines

B. Concurrent Requirements

None.

C. <u>Reason</u>

(1) Problem

A change in the coating application process from High Energy Plasma to Standard Energy Plasma, has meant that release of particles from the HP Compressor front stubshaft tungsten carbide cobalt hard coating may occur. This may result in surface spall of the bearing race, which may lead to premature deterioration of the number 3 bearing.

The purpose of this Service Bulletin is to advise Operators of the availability of a new stubshaft part number, coated with High Energy Plasma.

(2) Evidence

The problem has been experienced on engines in service.

(3) Substantiation

The changes introduced by this modification have been the subject of extensive engineering analysis.

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(4) Objective

Incorporation of this modification is designed to maintain engine reliability.

- (5) Effect of Bulletin on:
 - (a) Operation

Not affected.

(b) Maintenance

Affected (see 1.0. Other Publications Affected).

(c) Overhaul

Not affected.

(d) Repair Schemes

None.

(e) Interchangeability

Not affected.

(f) Fits and Clearances

Not affected.

D. <u>Description</u>

- (1) The changes introduced are:
 - (a) A revised HP Compressor front stub shaft which is similar to the existing stub shaft except for the following:
 - (i) The option to use the standard plasma application process for the tungsten carbide cobalt coating has been deleted.
 - (ii) The requirement to use fluorescent penetrant inspection of the coating after grinding has been introduced.
 - (iii) The classification of the part has been changed from 'sensitive' to 'sensitive, traceable'.



E. Compliance

Category 7

Accomplish when the supply of superseded parts has been depleted.

F. Approval

The part number transactions shown in section 2. Material Information of this Service Bulletin have been shown to comply with the applicable Federal Aviation Regulations and are FAA approved for the engine models listed.

G. Manpower

(1) In service

Not applicable.

(2) At overhaul

Rework only - 10 hours.

<u>NOTE</u>: The parts affected by this Service Bulletin are accessible at overhaul.

H. Material Price and Availability

Modification kit not required; parts supplied as single line items.

I. Tooling Price and Availability

Special tools are not required.

J. <u>Industry Support Information</u>

None.

K. Weight and Balance

(1) Weight Change

None.

(2) Moment Arm

Not affected.

(3) Datum

Engine front mount centreline (Power Plant Station (PPS) 100).

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L. Electrical Load Data

The aircraft electrical load is not affected by this Service Bulletin.

M. Software Accomplishment Summary

Not applicable.

N. References

- R (1) IAE Engineering Change Numbers O2VROO1, O3VR761B.R, O5VR915 and O5VR915A.
- R (2) Quality Report V2500/11/03-0011.
 - (3) ATA Locator 72-32-23.
 - (4) Engine Manual, 72-00-40, Removal/Installation.

O. Other Publications Affected

- (1) Illustrated Parts Catalogue (IPC) 2IA, 2IB, 3IA, 3IB, 5IA, 5IB, 6IA, 6IB, 7IA and 7IB, Chapter/Section 72-32-23, will be revised.
- (2) Engine Manual, 72-32-23, Config-01, Cleaning-02 and Inspection/Check-02.

P. Interchangeability of Parts

Not affected.



2. Material Information

A. The kit required consists of the following parts:

None.

B. Parts to be reworked:

All Engines

72-32-23

FIG ITEM NO.	NEW PART NO.	QTY	PART TITLE	MAT	OLD PART NO.	INSTR DISP
01700	6A8093	1	.Shaft and Seal Assembly Front, HP Compressor		6A4468	(1D)(S1)
01745	6A8094	1	Shaft, Balancing and Assembly - Front, HP Compressor	-	6A4147	(1D)(S1)
01770	6A8095	Ref	Shaft - Front, HP Compressor	-	6A4148	(B)(1D) (S1)

C. New production parts:

All Engines

72-32-23

FIG ITEM NO.	NEW PART NO.	QTY	PART TITLE	MAT	OLD PART NO.	INSTR DISP
01700	6A8082	1	.Shaft and Seal Assembly Front, HP Compressor		6A4468	(A)(S1)
01745	6A8083	1	Shaft, Balancing and Assembly - Front, HP Compressor	-	6A4147	(A)(S1)
01770	6A8084	Ref	Shaft - Front, HP Compressor	-	6A4148	(C)(S1)

D. <u>Instructions disposition codes:</u>

- (A) New part will be made available from August 2002.
- (B) Part of 6A8094.
- (C) Part of 6A8083.

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- (S1) New and old parts are freely and fully interchangeable.
- (1D) Old part may be reworked and re-identified to the new part number.

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3. Accomplishment Instructions

A. Prepare the HP Compressor

Remove the HP Compressor Assembly from the engine (72-00-41). Disassemble the HP Compressor Assembly and the HP Compressor Rotor (72-00-41). Remove Number 3 Bearing internal gearbox and support assembly (72-32-00), Disassemble the Number 3 Bearing and Bevel Gear (72-32-20). Disassemble the Seal Ring Carrier (72-32-23) to remove the HPC Front Stubshaft. Refer to the applicable Engine Manual (EM), Removal/Installation.

B. Rework Instructions

(1) This is a Source Demonstrated Service Bulletin which can only be done by approved vendors.

The Source Demonstration requirements of this Service Bulletin means that any facility not authorised to accomplish this Service Bulletin either utilize the Authorised Repair Vendors listed below or contact the IAE Repair Services Group to determine if a qualification programme can be initiated at their facility.

IAE - International Aero Engines AG

c/o Rolls-Royce plc.

P0 Box 31

DERBY

DE24 8BJ

ENGLAND

For Attention: Manager - Technical Services

Authorised Repair Vendors for Service Bulletin 72-0421 are listed below:

Rolls-Royce

Aero Repair and Overhaul

ANSTY

COVENTRY

CV7 9JR

ENGLAND

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Ishikawajima - Harima Heavy Industries Co., Ltd. (IHI)

229 Tonogaya Mizuho-machi,

Nishitama-gun,

TOKYO 190 - 1297

JAPAN

Attn: Customer Support Manager, Maintenance Department, Civil Aero-Engine Division, Aero-Engine and Space Operations

Lufthansa Technik AG

Ham WP

P0 Box 63 03 00

D-22313 HAMBURG AIRPORT

GERMANY

The designation by IAE of an Authorized Repair Vendor indicates that the Repair Vendor has demonstrated the necessary capability to enable it to carry out the listed Service Bulletin Work. However, IAE makes no warranties or representations concerning the qualifications or quality standards of the Repair Vendors to carry out the Service Bulletin work, and accepts no responsibility whatsoever for any work that may be carried out by a Repair Vendor other than when IAE is listed as the Repair Vendor. Authorized Repair Vendors do not act as agents or representatives of IAE.

(2) Standard Equipment:

Refer to approved vendor process.

(3) Consumable materials:

Refer to approved vendor process.



(4) Rework the following parts:

6A4147 - HP Compressor Front Stubshaft. Refer to Illustrated Parts Catalogue (IPC) Chapter/section 72-32-23, Fig/Item 01-745.

NOTE: Part serial numbers contained within Appendix A do not require this rework (as they have previously had the high energy coating applied) and should be re-identified by vibro-engraving the number 10 within a triangle adjacent to the existing identity. Refer to SPM TASK 70-09-00-400-001. Use vibro-engraving equipment.

<u>CAUTION</u>: TITANIUM COMPONENT - YOU MUST USE SILICON CARBIDE TYPE ABRASIVE WHEELS, STONES AND PAPERS TO DRESS, BLEND AND POLISH THIS

COMPONENT.

<u>CAUTION</u>: TITANIUM COMPONENT - AVOID BUILD UP OF HEAT BY APPLYING ONLY GENTLE PRESSURE AND KEEPING THE TOOL SPEED AS LOW AS POSSIBLE.

CAUTION: TITANIUM COMPONENT - YOU MUST MAKE SURE THAT WHEN YOU DRESS
MATERIAL, BLEND AND POLISH, TO MAKE SMOOTH, THAT NO SPARKS ARE

PRODUCED.

CAUTION: TITANIUM COMPONENT - IF THE MATERIAL SHOWS A CHANGE IN COLOUR,

TO DARKER THAN A LIGHT STRAW COLOUR, THE COMPONENT IS TO BE

REJECTED.

PROCEDURE RELATED DATA

(5) Remove the hard coating Refer to Figure 1
Refer to approved vendor process.

(6) Measure the surface(s) to be Refer to approved vendor process. Use dimensional inspection equipment.

(7) Do a swab etch

(a) Chemically clean Refer to SPM TASK 70-11-08-300-503.

Use chemical cleaning equipment.

(b) Swab etch the area(s) of Refer to SPM TASK 70-11-08-300-503.

repair Use chemical cleaning equipment.

(8) Do a penetrant crack test Refer to approved vendor process.

(9) Chemically clean the Front Refer to SPM TASK 70-11-26-300-503.

Stubshaft Use chemical cleaning equipment.

(10) Mask area(s) not to be repaired Refer to approved vendor process.

(11) Abrasive blast the surface(s) to be Refer to approved vendor process.

repaired

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(12) Metal spray the surface(s) for Refer to approved vendor process. repair (13) Grind the hard coating on the HP Refer to Figures 1 and 2. Compressor Front Stubshaft Refer to approved vendor process. (14) Visually and dimensionally inspect Refer to Figures 1, 2 and 3. Use dimensional inspection equipment. (15) Do a penetrant crack test Refer to approved vendor process. Refer to SPM TASK 70-09-00-400-001. (16) Cancel the existing part number and

re-identify the HP Compressor Front Stubshaft Assembly with the new part number adjacent to the existing part number

Use vibro-engraving equipment.

Existing Re-number

6A4147 6A8094

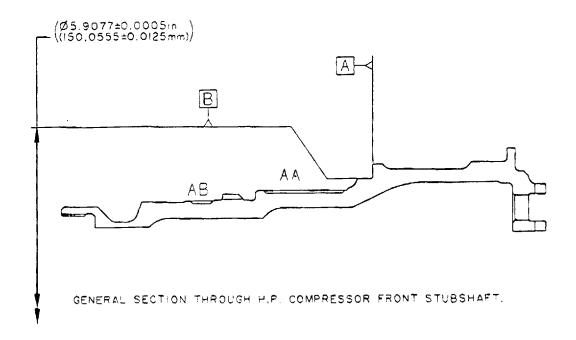
C. Assembly Instructions

Re-Assemble the HPC Front Stubshaft and the Seal Ring Carrier (72-32-23), Number 3 Bearing and Bevel Gear (72-32-20), Install Number 3 Bearing Internal Gearbox and Support Assembly (72-32-00), Assemble the HP Compressor Assembly (72-41-00) and Install the HP Compressor (72-00-41). For all other assembly instructions refer to Engine Manual, 72-00-40, Removal/Installation.

D. Recording Instructions

A record of accomplishment is necessary.





GEOMETRIC SYMBOLS CONFORM TO ISO RIOIL SURFACE TEXTURE IN MICROINCHES (MICROMETRES)

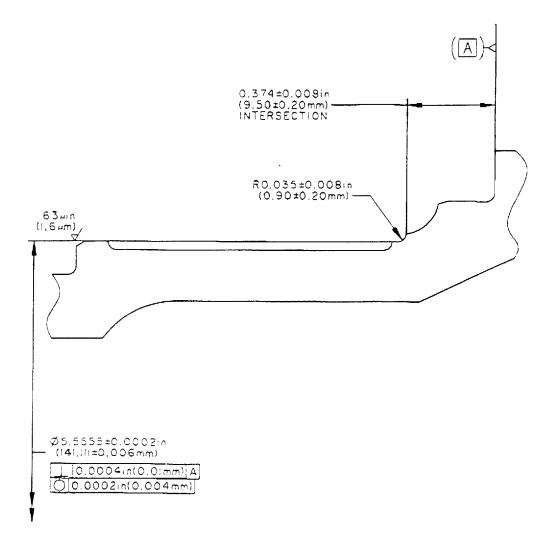
MACHINE WHERE SHOWN \$\notin{align*}{2}\$.

APPLY THE FLAME DEPOSITION MATERIAL TO AREAS SPECIFIED AA AND AB.

Ted0004528

General section through HP Compressor Front Stubshaft Figure 1

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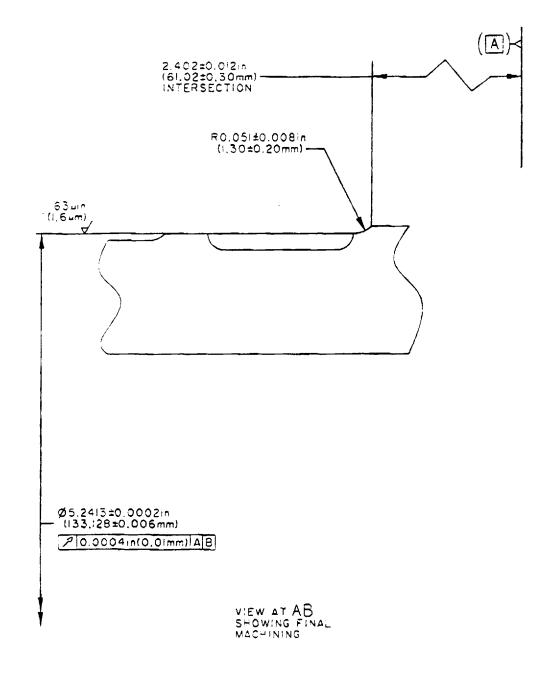
VIEW AT AA SHOWING FINAL MACHINING

Rework of HP Compressor Front Stubshaft Figure 2

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Tded0004529





Ted0004530

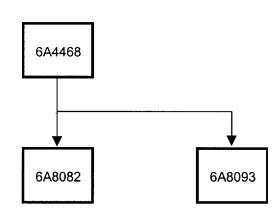
Rework of HP Compressor Front Stubshaft Figure 3

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HP Compressor Front Stub Shaft Assy

Baseline

V2500-ENG-72-0421
Engine - HP Compressor Front Stub
Shaft - Introduction of a Revised HPC
Front Stub Shaft with High Plasma
Coating.



ded00043

Family Tree - HP Compressor Front Stubshaft Assembly

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R

SERVICE BULLETIN

APPENDIX A

R <u>HP Compressor Stubshafts plasma sprayed to RPS 629 (High Energy) standard at new production</u>

R R7BFGQ, R7BFGR, R7BFGS, R7BFGT, R7BFGU, R7BFGV, R7BFHF, R7BFHG, R7BFHH, R7BGCY, R R7BGGD, R7BGKN, R7BGKS, R7BGKT, R7BGKU, R7BGKV, R7BGKW, R7BGKX, R7BGKY, R R7BGKZ, R7BGLA, R7BGLB, R7BGLC, R7BGLD, R7BGLE, R7BGLF, R7BGLG, R7BGLH, R7BGLJ, R R7BGLK, R7BGLL, R7BGLN, R7BGLP, R7BGLQ, R7BGLR, R7BGLS, R7BGLT, R7BGLU, R R7BGLV, R7BGLW, R7BGLX, R7BGLY, R7BGLZ, R7BGMA, R7BGMB, R7BGMC, R7BGMD, R7BGME, R R7BGMF, R7BHFT, R7BHFU, R7BHFV, R7BHFW, R7BHFX, R7BHFY, R7BHFZ, R7BHGG, R7BHGH, R R7BHGJ, R7BHGL, R7BHGM, R7BHGO, R7BHGP, R7BHGQ, R7BHGR, R7BHGS, R7BHGT, R R7BHJN, R7BHJR, R7BHJS, R7BHMF, R7BHNN, R7BHNP, R7BHNQ, R7BHNR, R7BHNT, R R7BHNU, R7BHNV, R7BHNW, R7BHNX, R7BHNY, R7BHSF, R7BHSG, R7BHSH, R7BHSJ, R7BHSK, R R7BHSL, R7BHSM, R7BHSN, R7BHSP, R7BHSQ, R7BHSR, R7BHSS, R7BHST, R7BHSU, R7BHTB, R R7BHTC, R7BHTD, R7BHTE, R7BHXU, R7BHXV, R7BHXW, R7BHXX, R7BHXY, R7BJAP, R7BJAQ, R R7BJAR, R7BJAS, R7BJAT, R7BJAU, R7BJAV, R7BJAW, R7BJAX, R7BJAY, R7BJAZ, R7BJBA, R R7BJBB, R7BJBC, R7BJBD, R7BJBE, R7BJBF, R7BJBG, R7BJBH, R7BJBJ, R7BJBK, R7BJBL, R R7BJBM, R7BJBN, R7BJBP, R7BJBQ, R7BJBR, R7BJBS, R7BJBT, R7BJBU, R7BJBV, R7BJHX, R R7BJHY, R7BJHZ, R7BJJA, R7BJJB, R7BJJC, R7BJJD, R7BJJE, R7BJJF, R7BJJG, R7BJJH, R R7BJJJ, R7BJJK, R7BJJL, R7BJJM, R7BJJN, R7BJJP, R7BLTD, R7BLTE, R7BLTF, R7BLTG, R R7BLTH, R7BLTJ, R7BLTK, R7BLTL, R7BLTM, R7BLTN, R7BLTP, R7BLTQ, R7BLTR, R7BNHK, R R7BNHL, R7BNHM, R7BNHN, R7BNHP, R7BNHQ, R7BNHR, R7BNHS, R7BNHT, R7BNHU, R7BNHV, R R7BNHW, R7BNHX, R7BNHY, R7BNHZ, R7BNJA, R7BNJB, R7BNJC, R7BNJD, R7BNJE, R7BNJF, R R7BNJG, R7BNJH, R7BNJJ, R7BNJK, R7BNJL, R7BNJM, R7BNJN, R7BPMA, R7BPMB, R7BPMC, R R7BPMD, R7BPME, R7BPMF, R7BPMG, R7BPMH, R7BPMJ, R7BPMK, R7BPML, R7BPMM, R7BPMN, R R7BPMP, R7BPMQ, R7BPMR, R7BPMS, R7BPMT, R7BPMU, R7BPMV, R7BPMW, R7BPMX, R7BPMY, R R7BPMZ, R7BPNF, R7BPNG, R7BPNH, R7BPNJ, R7BPNK, R7BPNL, R7BPNM, R7BPNP, R7BQFR, R R7BQFS, R7BQFT, R7BQFU, R7BQFV, R7BQFW, R7BQFX, R7BQFY, R7BQFZ, R7BQGA, R7BQGB, R R7BQGC, R7BQGD, R7BQNL, R7BQNL, R7BQNM, R7BQNN, R7BQNP, R7BQNQ, R7BQNR, R7BQNS, R R7BQNT, R7BQNU, R7BQNV, R7BRFA, R7BRFB, R7BRFC, R7BRFD, R7BRFE, R7BRFF, R7BRFG, R R7BRFH, R7BRFJ, R7BRFK, R7BRFL, R7BRFM, R7BRFN, R7BRFR, R7BRFS, R7BRFT, R7BRPC, R R7BRPD, R7BRPE, R7BRPF, R7BRPG, R7BRPH, R7BRPJ, R7BRPK, R7BRPL, R7BRPM, R7BRPN, R R7BRPP, R7BRPR, R7BRPU, R7BRPV, R7BSKX, R7BSKY, R7BSKZ, R7BSLA, R7BSLB, R7BSLC, R R7BSLD, R7BSLE, R7BSLF, R7BSLG, R7BSLH, R7BSLJ, R7BSLK, R7BSLL, R7BSLM, R7BSLN, R R7BSLP, R7BSLQ, R7BSLR, R7BSLS, R7BSLT, R7BSLU, R7BSLV, R7BSLW, R7BSLX, R7BSLY, R R7BSLZ, R7BSMA, R7BSMB, R7BSMC, R7BSMD, R7BSME, R7BSMF, R7BSMG, R7BSMH, R7BSMJ, R R7BSMK, R7BSML, R7BSMM, R7BSMN, R7BTTJ, R7BTVG, R7BUJN, R7BUJP, R7BUJQ, R7BUJR, R R7BUJS, R7BUJT, R7BUJU, R7BUJV, R7BUJW, R7BUJX, R7BUJY, R7BUJZ, R7BUKA, R7BUKB, R R7BUKC, R7BUKD, R7BUKE, R7BUKF, R7BUKG, R7BUKH, R7BUKJ, R7BUKK, R7BUKL, R7BUKM, R R7BUKN, R7BUKP, R7BUKQ, R7BUKR, R7BUKS, R7BUKT, R7BV5Q, R7BV5V, RYBVBH, R7BVBK, R R7BVBL, R7BVBM, R7BVBN, R7BVSB, R7BVSC, R7BVSD, R7BVSE, R7BVSF, R7BVSG, R7BVSP, R R7BVSQ, R7BVSR, R7BVSS, R7BVST, R7BVSU, R7BVSV, R7BVSW, R7BVSY, R7BVSZ, R7BVTA, R R7BVTB, R7BVTC, R7BVTD, R7BVTG, R7BVTH, R7BXEY, R7BXEZ, R7BXFA, R7BXFC, R7BXFD, R R7BXFE, R7BXFF, R7BXFG, R7BXFH, R7BXFJ, R7BXFK, R7BXFL, R7BXFM, R7BXFN, R7BXFP, R R7BXFQ, R7BXFR, R7BXFS, R7BXFT, R7BYFL, R7BYTZ, R7BYUA, R7BYUB, R7BYUC, R7BYUD,

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Appendix A - Page 1 of 3

R R7BYUE, R7BYUF, R7BYUG, R7BYUH, R7BYUJ, R7BYUK, R7BYUL, R7BYUM, R7BYUN, R7BYUP,

```
R R7BYUQ, R7BYUR, R7BYUS, R7BYUT, R7BYUU, R7BYUW, R7BYUW, R7BYUX, R7BYUY, R7BYUZ,
R R7BYVA, R7BYVB, R7BYVC, R7BYVD, R7BYVE, R7BYVF, R7BYVG, R7BYVP, R7BYVQ, R7BYVR,
R R7BZAG, R7BZAJ, R7BZAK, R7BZAL, R7BZAM, R7BZAN, R7BZAP, R7BZAQ, R7BZAR, R7BZAS,
R R7BZAT, R7BZAU, R7BZAV, R7BZAW, R7BZAX, R7BZAY, R7BZAZ, R7BZBA, R7BZBB, R7BZBC,
R R7BZBD, R7BZBE, R7BZBF, R7BZBG, R7BZBH, R7BZBJ, R7BZBK, R7BZBL, R7BZBM, R7BZBN,
R R7BZBP, R7BZBR, R7BZBS, R7BZBT, R7BZBU, R7BZBV, R7BZBW, R7BZBX, R7BZBY,
R R7BZBZ, R7BZCA, R7BZCB, R7BZCC, RYBZCD, R7BZCE, R7BZCF, R7BZCG, RYBZCH, R7CBKR,
R R7CBKS, R7CBKT, R7CBKU, R7CBKV, R7CBKW, R7CBKX, R7CBKY, R7CBKZ, R7CBLA, R7CBLB,
R R7CBLC, R7CBLD, R7CBLE, R7CBLF, R7CBLG, R7CBLH, R7CBLJ, R7CBLK, R7CBLL, R7CBLM,
R R7CBLN, R7CBLP, R7CBLQ, R7CBLR, R7CBLS, R7CBLT, R7CBLU, R7CBLV, R7CBLW, R7CBLX,
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<u>ENGINE - HP COMPRESSOR FRONT STUBSHAFT - INTRODUCTION OF A REVISED HP COMPRESSOR</u> <u>FRONT STUBSHAFT WITH HIGH ENERGY PLASMA COATING</u>

SUPPLEMENT - PRICES AND AVAILABILITY

The prices (if shown) are for estimating purposes only and as such are given in good faith, without commercial liability for advanced planning purposes only. Refer to IAE Spares and/or current price catalogue for current prices.

1. Modification Kit:

Not applicable.

2. Parts required:

Part No.	Description	Unit Price US Dollars
6A8082	Shaft and Seal Assy - Front, HP Compressor	35,090.00
6A8083	Shaft, Balancing and Assy - Front, HP Compressor	30,149.00

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