SERVICE BULLETIN

Date: July 24, 1997

Subject: Transmittal of Revision 1 to Service Bulletin Number V2500-NAC-78-0103

Service Bulletin Revision History:

Event

Date

Basic Issue

May. 08/97

Revision 1

Jul. 24/97

Reasons for Issuance of Revision

(1) To add a note stating this Service Bulletin is issued for parts progression purposes only and has been superseded by the content of Service Bulletin V2500-NAC-78-0114.

Effect on Past Compliance

(1) None.

List of Effective Pages:

Page No	<u>) .</u>	Rev. No.	<u>Date</u>	
1 thru 4 thru	_	1 Basic		24/97 08/97

V2500-NAC-78-0103

Transmittal Page 1 of 1



V2500 Propulsion System - Nacelle

International Aero Engines SERVICE BULLETIN

NACELLE - EXHAUST - BOLT, PRESSURE RELIEF DOOR LANYARD -REPLACEMENT OF

MODEL APPLICATION

V2500-D5

BULLETIN INDEX LOCATOR

78-32-00

Compliance Category Code

6

Internal Reference No. LL/AC 95VN926

May 8, 1997 Revision 1 - July 24, 1997 V2500-NAC-78-0103 Page 1 of 8 R

V2500 Propulsion System — Nacelle

Aero Engines SERVICE BULLETIN

1. Planning Information

NOTE: This Service Bulletin is issued for parts progression history purposes only. The change introduced by this Service Bulletin has been superseded by the change introduced by Service Bulletin V2500-NAC-78-0114.

A. Effectivity

- (1) Aircraft: MD90
- (2) Nacelle: V2500-D5 thrust reverser serial numbers 0021001 through 0044001.

B. Reason

(1) Condition

Due to possible clamp up condition of lanyard, it may be difficult to fully open pressure relief door.

(2) Background

Insufficient grip length of bolt.

(3) Objective

Eliminate possible undesirable clamp up of pressure relief door lanyards.

(4) Substantiation

Trial fitment of new bolt and laminated washer has proven successful.

(5) Impact of Bulletin on Workshop Procedures:

Removal/Installation Not Affected Disassembly/Assembly Not Affected Cleaning Not Affected Inspection/Repair Not Affected

May 8, 1997 Revision 1 - July 24, 1997 V2500-NAC-78-0103



V2500 Propulsion System — Nacelle

Aero Engines SERVICE BULLETIN

Repair Testing Not Affected Not Affected

(6) Supplemental Information

None.

C. Description

The change introduced by this Bulletin is to replace the existing bolt with a longer grip standard bolt and add a laminated washer.

D. Approval

Incorporation of this Service Bulletin must be accomplished only in conjunction with Douglas Aircraft Company Service Bulletin MD-90-78-020 which has received exclusive FAA approval for MD-90 Series aircraft.

E. Compliance

Category 6

Accomplish when the nacelle subassembly (i.e., accessories, componenets) is disassembled sufficiently to afford access to the affected part and to all affected spare parts.

F. Manpower

Estimated manhours per nacelle to incorporate the full intent of this Bulletin:

VENUE

ESTIMATED MANHOURS

(1) In Service

Not Applicable

(2) At overhaul

(a) To rework

0.5 M/Hrs. 0.5 M/Hrs.

May 8, 1997 Revision 1 - July 24, 1997 V2500-NAC-78-0103

V2500 Propulsion System — Nacelle

Aero Engines SERVICE BULLETIN

NOTE: After incorporation of this modification, a maximum of 1.0 manhours for labor will be reimbursed by Rohr, as a labor credit allowance per affected aircraft to obtain a labor credit allowance after procurement of noted material. Labor claims should reference this service bulletin number and aircraft fuselage number and be submitted to:

Rohr, Inc.
850 Lagoon Drive
Chula Vista, CA. 91910-2098
U.S.A.
Attn: Airline Support Manager, MZ 107A
Warranty Department
(Ref. Service Bulletin V2500-NAC-78-0103)

G. Material Cost and Availability

Modification kit not required. Parts are supplied as single line items (Refer to Section 3, Material Information).

Operators with units listed in Paragraph 1.A. should submit a no-charge purchase order for the applicable quantity of parts. The purchase order must reference this service bulletin number with applicable Operator serial numbers and list only the parts shown. Operators will have one year from issue date of the Service Bulletin to place an order. After one year, Operators will have to oder parts at catalog price if they desire to incorporate the change.

Direct Purchase Order to:
Rohr Inc.
850 Lagoon Drive
Chula Vista, CA. 91910-2098
U.S.A.
Attn: Airline Support Manager, MZ 107A
(Ref. Service Bulletin V2500-NAC-78-0103)

V2500-NAC-78-0103

Page 4

May 8, 1997



V2500 Propulsion System — Nacelle

Aero Engines SERVICE BULLETIN

H. Tooling Cost and Availability Not applicable.

- I. Weight and Balance
 - (1) Weight change No effect
 - (2) Moment arm No effect
 - (3) Datum Front Engine Mount Centerline (Power Plant Station (PPS) 95.1)
- J. Electrical Load Data

Not affected.

K. References Chapter/Section
IAE V2500 Standard Practices/Processes 70-09-00
Manual (SPP-V2500-1IA)

L. Other Publications Affected

V2500/MD90 Engine Illustrated 78-32-09 Parts Catalog (S-V2500-3IA)

V2500/MD90 Thrust Reverser Component 78-32-09 Maintenance Manual (CMM-TR-V2500-3IA)

V2500-NAC-78-0103

Page 5

May 8, 1997

Not subject to the EAR per 15 C.F.R. Chapter 1, Part 734.3(b)(3).

Aero Engines SERVICE BULLETIN

2. Accomplishment Instructions

A. Pre-requisite Instructions

None.

- B. Rework or Modification Instructions
 - (1) Remove four NAS1291C4M nuts and four 290-0889-501 bolts from four lanyards and two terminal fork ends on pressure relief door. Refer to Figure 1. Discard bolts. Keep nuts.
 - (2) Install four NAS6304U10 bolts and four NAS1291C4M nuts on four lanyards and two terminal fork ends on thrust reverser pressure relief door. Torque nut to 10-20 lbf-in (1.1-2.4 Nm).
 - (3) Measure gaps between nuts and lanyards. Gap should be 0.005-0.015 in (0.13-0.38 mm).
 - (4) If any gaps are greater than 0.015 inch (0.38 mm), remove applicable NAS1291C4M nut. Install necessary thickness of S700W0029-6 laminated washer on NAS6304U10 bolt to obtain proper gap. Install NAS1291C4M nut and tighten to a torque of 10-20 lbf-in (1.1-2.4 Nm).
- C. Post-requisite Instructions

None.

D. Recording Instructions

A record of accomplishment is necessary. Write in applicable records and metal stamp, vibroetch, or electroetch on thrust reverser data plate that Service Bulletin V2500-NAC-78-0103 has been done. Refer to IAE V2500 Standard Practices/Processes Manual, Chapter 70-09-00.

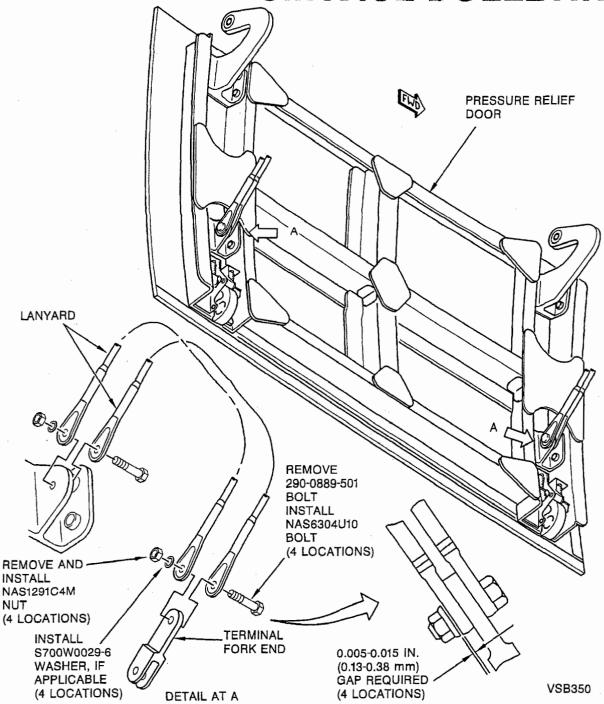
V2500-NAC-78-0103

Page 6

May 8, 1997



International Aero Engines SERVICE BULLETIN



PRESSURE RELIEF DOOR LANYARD BOLT REPLACEMENT FIGURE 1

May 8, 1997

V2500-NAC-78-0103

Aero Engines SERVICE BULLETIN

3. Material Information

Applicability: For each V2500-D5 thrust reverser to incorporate this Bulletin.

A. Kits Associated with this Bulletin:

None.

B. Parts Affected by this Bulletin:

NEW PART NO. (ATA NO.)	OTY	EST'D UNIT PRICE	KEYWORD	OLD PART NO. (IPC NO.)	INSTR/ DISPOS
NAS6304U10	4		Bolt	290-0889-501	(A)(1D)
(78-32-09)			*	(01-065)	(S1)
S700W0029-6	4	i i	Washer		(A)(S1)
(78-32-09)				(01-063)	

- C. Instruction/Disposition Code Statements
 - (A) New part is currently available.
 - (1D) Discard old part.
 - (S1) New parts coded (S1) are interchangeable in complete sets.

NOTE: The estimated 1997 unit prices shown are provided for planning purposes only and do not constitute a firm quotation. Consult the Rohr Price Catalog or contact Rohr's Customer Support Department for information concerning firm prices.

D. Materials Required to Incorporate This Bulletin:

None.

V2500-NAC-78-0103